## ENCE 353 Final Exam, Open Notes and Open Book

Name:

Exam Format and Grading. This take home final exam is open notes and open book. You need to comply with the university regulations for academic integrity.

Answer Question 1. Then answer three of the five remaining questions. Partial credit will be given for partially correct answers, so please show all your working.

IMPORTANT: Only the first four questions that you answer will be graded, so please cross out the two questions you do not want graded in the table below. Also, before submitting your exam, check that every page has been scanned correctly.

| Question | Points | Score |
| :---: | :---: | :---: |
| 1 | 20 |  |
| 2 | 10 |  |
| 3 | 10 |  |
| 4 | 10 |  |
| 5 | 10 |  |
| 6 | 10 |  |
| Total | 50 |  |

Question 1: 20 points

COMPULSORY: Computing Displacements with the Method of Virtual Forces. Figure 1 is a front elevation view of a cantilevered beam carrying a single point load P. EI is constant along the beam structure A-B-C-D.


Figure 1: Cantilevered beam carrying a single applied load P .
[1a] (3 pts) Use the method of virtual displacements to compute the support reactions at B and C.
[1b] (2 pts) Draw and label the M/EI diagram.
[1c] (5 pts) Use the method of virtual forces to show that the vertical deflection (measured downwards) at D is:

$$
\begin{equation*}
y_{D}=\frac{P a^{2}}{3 E I}(a+b) . \tag{1}
\end{equation*}
$$

Show all of your working.
[1d] (5 pts) Use the method of virtual forces to show that the vertical deflection (measured downwards) at A is:

$$
\begin{equation*}
y_{A}=\frac{P a^{2} b}{6 E I} \tag{2}
\end{equation*}
$$

Show all of your working.
[1e] (5 pts) Use the method of virtual forces to show that rotation of the beam (measured clockwise) at D is:

$$
\begin{equation*}
\theta_{D}=\frac{P a}{6 E I}[2 b+3 a] \tag{3}
\end{equation*}
$$

Show all of your working.

## Question 2: 10 points

OPTIONAL: Derive Elastic Curve for Beam Deflection. Figure 2 is a front elevation view of a cantilevered beam carrying a single point load P. EI is constant along the beam structure A-B-C-D.


Figure 2: Cantilevered beam carrying a single applied load P .
[2a] (5 pts) Starting from the differential equation,

$$
\begin{equation*}
\frac{d^{2} y}{d x^{2}}=\left[\frac{-M(x)}{E I}\right] \tag{4}
\end{equation*}
$$

(notice the minus sign on $\mathrm{M}(\mathrm{x})$ ) and appropriate boundary conditions, show that that vertical beam deflection along the beam segment B-C is:

$$
\begin{equation*}
y(x)=\frac{P a}{6 E I b}\left(b^{2}-x^{2}\right) x . \tag{5}
\end{equation*}
$$

[2b] (5 pts) Hence (i.e., using your results from part [2a] as a starting point), show that the maximum upwards deflection of the beam occurs at a distance $b / \sqrt{3}$ from $B$, and that its value is:

$$
\begin{equation*}
y_{\text {maximum upward }}=\frac{P a b^{2}}{9 \sqrt{3} E I} \tag{6}
\end{equation*}
$$

Show all of your working.

Question 3: 10 points

OPTIONAL: Principle of Superposition and Method of Moment Area. Figure 3 is a front elevation view of a multi-span beam structure that carries uniform load $w_{o}(\mathrm{~N} / \mathrm{m})$ along beam segment $\mathrm{B}-\mathrm{C}$. EI is constant along the beam structure A-B-C-D.


Figure 3: Multi-span beam carrying a uniform load on segment B-C.
[3a] (2 pts) Compute the degree of indeterminacy for this structure.
[3b] (2 pts) Draw and label a diagram indicating how the principle of superposition can be used to simplify the analysis of the multi-span beam structure.
[3c] (6 pts) Using the method of moment area, or otherwise compute the support reactions at supports A, B and C. Check that you reaction forces are in equilibrium with the applied loads.

Question 4: 10 points
OPTIONAL: Structural Analysis with Method of Virtual Displacements. The cantilevered beam structure shown in Figure 4 supports a uniform load distributed $\mathrm{w}(\mathrm{x})=w_{o}(\mathrm{~N} / \mathrm{m})$ over the beam section B-C-D.


Figure 4: Front elevation view of a simple beam structure.
[4a] (4 pts) Use the method of virtual displacements to compute formulae for the vertical reactions at A and C. Show all of your working.
[4b] (6 pts) Use the method of virtual displacements to compute a formula for the bending moment at $B$. Show all of your working.

Question 5: 10 points
OPTIONAL: Use Principle of Virtual Work to Compute Displacements. Consider the articulated cantilever beam structure shown in Figure 5.


Figure 5: Elevation view of articulated cantilever beam structure.

At Point A, the cantilever is fully fixed (no movement) to a wall. Point B is a hinge. Both members have cross section properties EI. A single point load $\mathbf{P}(\mathrm{N})$ is applied at node D as shown in the figure.
[5a] (2 pts). Draw and label the bending moment diagram for this problem.
[5b] (2 pts). Qualitatively sketch the deflected shape. Indicate regions of tension/compression, and any points where slope of the beam is discontinuous.
[5c] (6 pts). Use the method of virtual forces to compute the vertical displacement and end rotation of the beam at D .

Show all of your working.

Question [5c] continued:

Question 6: 10 points
OPTIONAL: Zero-Force Memebers, Virtual Work, and Flexibility Matrices. Consider the truss structure shown in Figure 6.


Figure 6: Elevation view of a pin-jointed truss.

The horizontal and vertical degrees of freedom are fully-fixed at supports A and G. The truss carries vertical loads $P_{b}$ and $P_{d}$ at nodes B and D, respectively. Frame members AE, EH and $\mathbf{G H}$ have members have cross section properties 2AE. Otherwise, the members have cross section properties AE.
[6a] (3 pts) Use the method of virtual forces to compute the vertical deflection at node B due to load $P_{b}$ alone (i.e., $P_{d}=0$ ).

Question [6a] continued:
[ $6 \mathbf{b}]$ (3 pts) Use the method of virtual forces to compute the vertical deflection at node D due to load $P_{d}$ alone (i.e., $P_{b}=0$ ).
[6c] (4 pts) Use the method of virtual forces to compute the two-by-two flexibility matrix connecting the vertical displacements at points B and D to applied loads $P_{b}$ and $P_{d}$, i.e., as a function of $P_{b}, P_{d}$, L and AE.

$$
\left[\begin{array}{c}
\triangle_{b}  \tag{7}\\
\triangle_{d}
\end{array}\right]=\left[\begin{array}{ll}
f_{11} & f_{12} \\
f_{21} & f_{22}
\end{array}\right]\left[\begin{array}{l}
P_{b} \\
P_{d}
\end{array}\right] .
$$

