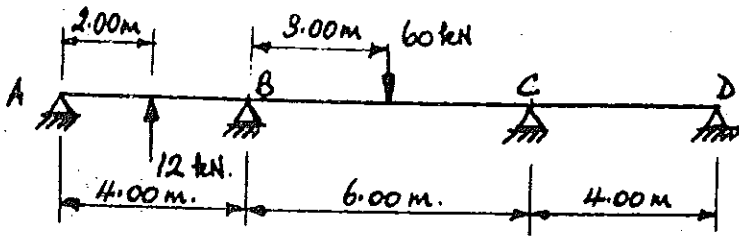
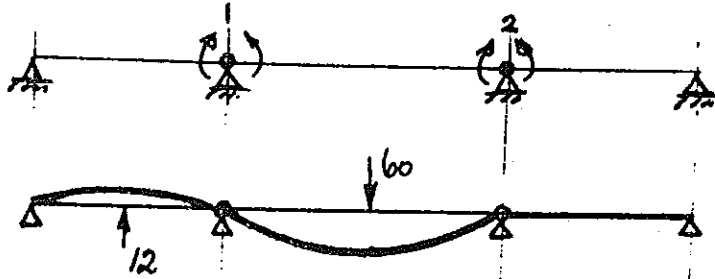


Example of a continuous beam analysed by the flexibility coefficient method.



The section stiffness of spans AB and CD is EI , and BC is $2EI$. Release the internal moments at supports B and C, and use point directions shown.

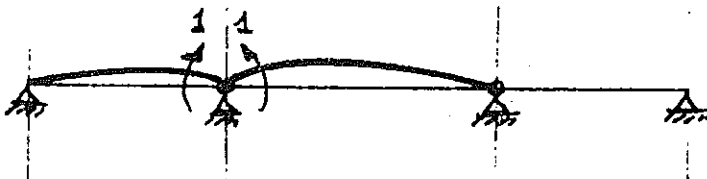


Rotations due to applied loads

Span AB, $\theta_B = \frac{12}{EI}$

Span BC, $\theta_B = -\frac{67.5}{EI}$; $\theta_C = -\frac{67.5}{EI}$

Total Rotations $\theta_B = -\frac{55.5}{EI}$; $\theta_C = -\frac{67.5}{EI}$



Rotations due to a unit moment at B

$f_{11} = \frac{2.33}{EI}$

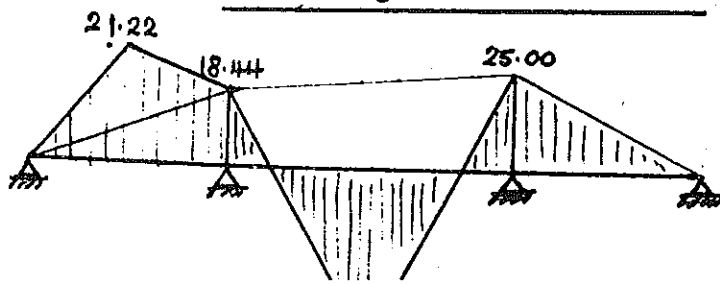
$f_{21} = \frac{0.50}{EI}$

Matrix relationship

$$-\begin{Bmatrix} -\frac{55.5}{EI} \\ -\frac{67.5}{EI} \end{Bmatrix} = \begin{bmatrix} \frac{2.33}{EI} & \frac{0.50}{EI} \\ \frac{0.50}{EI} & \frac{2.33}{EI} \end{bmatrix} \begin{Bmatrix} M_B \\ M_C \end{Bmatrix}$$

$\therefore M_B = 18.44 \text{ kN.m.}$

$M_C = 25.00 \text{ " "}$



Example (1). Temperature problem (suitable for lack-of-fit, shrinkage, creep)

A portal frame shown in Figure 1 is constructed from concrete ($E = 26000 \text{ Mpa}$) and has a constant cross section ($I = 700 \times 10^{-6} \text{ m}^4$). Obtain the distribution of bending moments introduced when the temperature of the left-hand column rises 10°C , the beam 20°C and the right-hand column 0°C . The coefficient of thermal expansion of concrete α is $10^{-5}/^\circ\text{C}$.

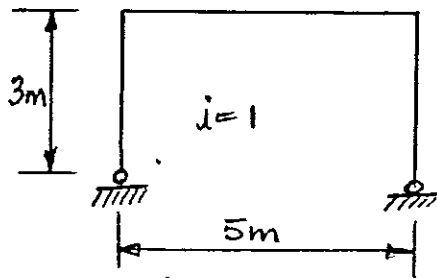
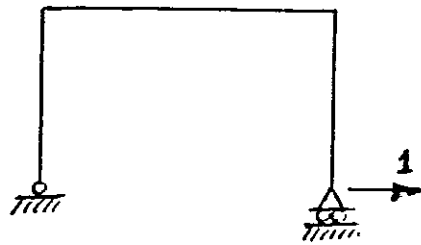
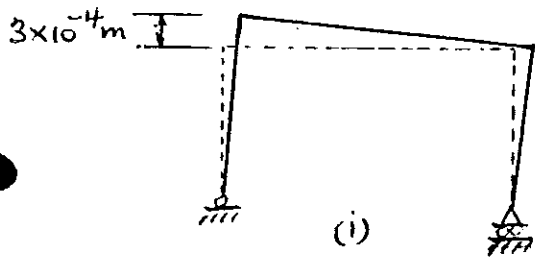
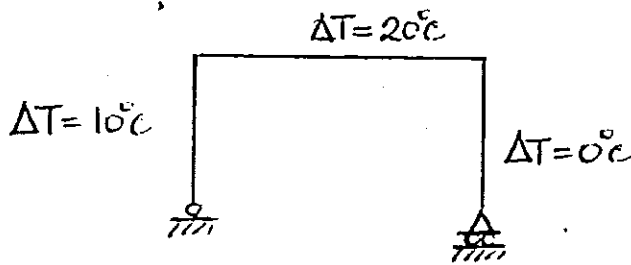


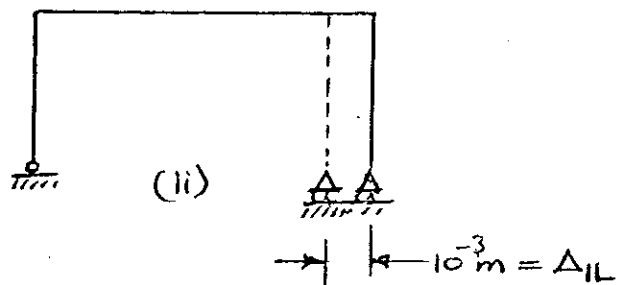
Figure 1



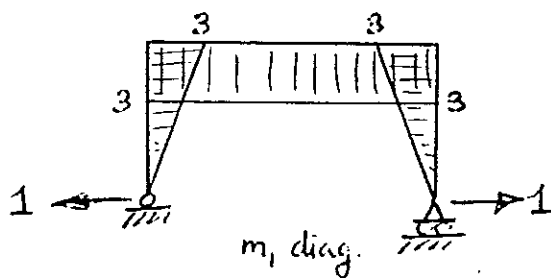
Point direction for horizontal redundant reaction release.



Expansion of
 (i) left-hand column = $10 \times 10^{-5} \times 3 = 3 \times 10^{-4} \text{ m}$
 (ii) beam = $20 \times 10^{-5} \times 5 = 10^{-3} \text{ m}$
 (iii) right-hand column = 0



Now obtain f_{11}

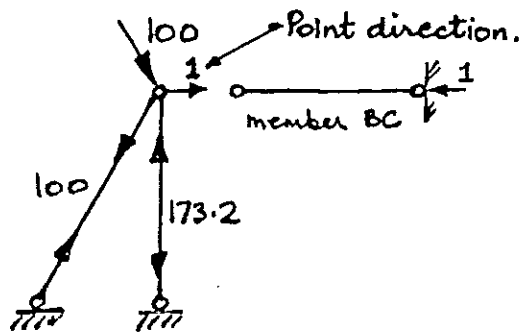
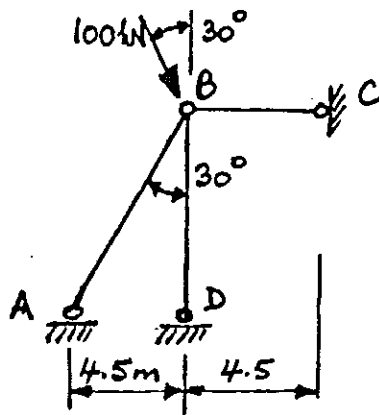


$$\begin{aligned}
 f_{11} &= \int_0^L \frac{m_1^2}{EI} ds \\
 &= \frac{2}{EI} \int_0^3 x^2 dx + \frac{1}{EI} \int_0^5 3^2 dx \\
 &= \frac{2}{EI} \left[\frac{x^3}{3} \right]_0^3 + \frac{9}{EI} [x]_0^5 \\
 &= \frac{63}{EI}
 \end{aligned}$$

$$f_{11} P_1 = -\Delta_{IL} \therefore P_1 = -0.0000158 EI = -288 \text{ N}$$

$$863 \times \frac{63}{EI} = 863 \text{ N.m.}$$

1) Member shortening.



$$E = 20 \text{ kN/mm}^2$$

$$A_{AB} = A_{BD} = 1000 \text{ mm}^2$$

$$A_{BC} = 1500 \text{ mm}^2$$

$$i = 6 + 3 - (2 \times 4) = 1$$

∴ Release one member force - eg. \overline{BC} .
Apply 100kN load and determine the deflection in point direction 1 of the primary structure

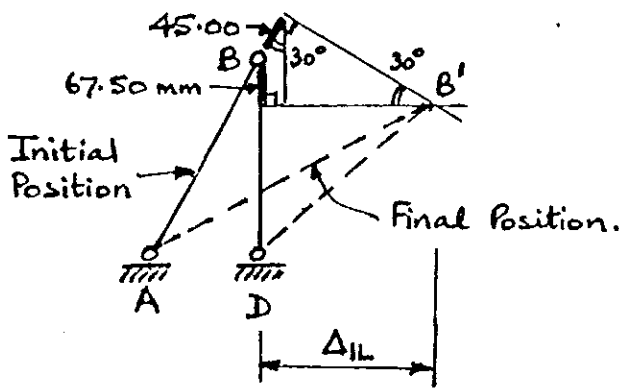
Shortening of BD

$$= \frac{173.2 \times \sqrt{3} \times 4500}{1000 \times 20} = 67.50 \text{ mm}$$

Lengthening of AB

$$= \frac{100 \times 2 \times 4500}{1000 \times 20} = 45.00 \text{ mm}$$

Displacement diagram :-



$$\Delta_{IL} = 45 \sin 30^\circ + \left(\frac{67.50 + 45 \cos 30^\circ}{\tan 30^\circ} \right)$$

$$\therefore \underline{\underline{\Delta_{IL} = 207.2 \text{ mm.}}}$$

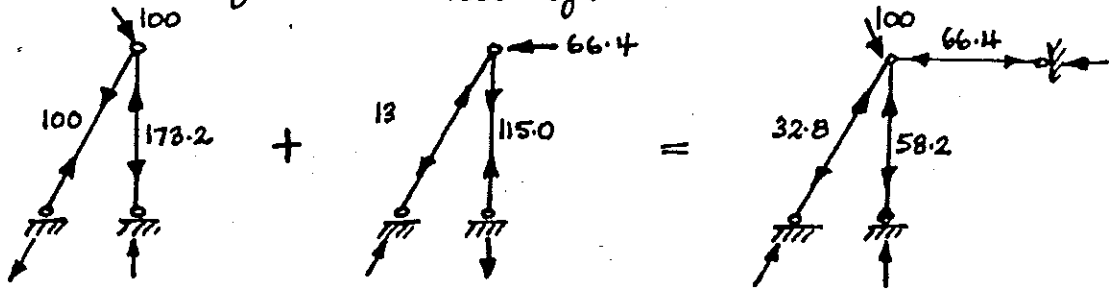
Shortening of the released member = $\frac{\overline{BC} \times 4500}{1500 \times 20} = 0.15 \overline{BC}$

The flexibility coefficient f_{11} has been determined in an earlier problem and is $13.2L/AE$ where L is the distance AD and AE are the member props. AB, BD . ∴ $f_{11} = \frac{13.2 \times 4500}{1000 \times 20} = 2.97$

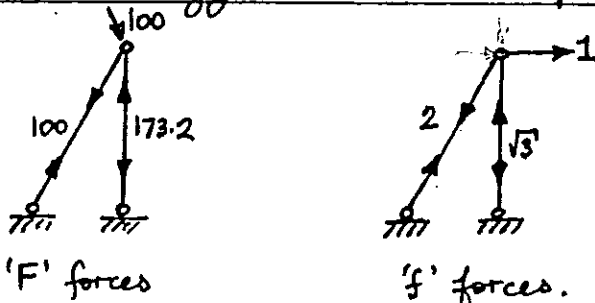
Eqn. compat.: $2.97 \overline{BC} + 0.15 \overline{BC} = -207.2$

$$\therefore \underline{\underline{\overline{BC} = -66.4 \text{ kN}}}$$

The final member forces consists of:-



Using the Energy theorem 'Principle of Virtual Forces'

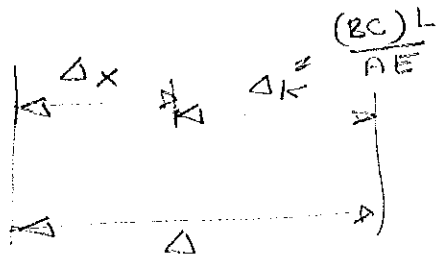


Member	L/AE	F	f	$\frac{FL}{AE}$	$\frac{fL}{AE}$	$f \cdot \bar{BC} + F$
AB	0.4500	100	2	90	1.800	-32.6
BD	0.3897	-173.2	$-\sqrt{3}$	116.9	1.169	-58.4
BC	0.1500	0	1	0	0.150	-66.3
				$\Sigma =$	206.9	3.119

Redundant Force $\bar{BC} = -\frac{206.9}{3.119} = -66.3 \text{ kN}$

$BC = -\frac{\Delta}{f_{11}}$

$f_{11} BC + \Delta = 0$



$\Delta_k = \frac{(\bar{BC})L}{AE}$

$-\Delta = f_{11} P_1 = \Delta$

$\Delta_x = \Delta - \Delta_k$
 $= f_{11} P_1 - BC \left(\frac{L}{AE}\right)$